DATA SEY CATHLOG # 15 MAGNETOMETER AND EPHEMERIS DATH

64-060H-00G 5 -02 A 5 C 8 D 8

stapes.

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### 1. INTRODUCTION:

The documentation for this data set was originally on paper, kept in NSSDC's Data Set Catalogs (DSCs). The paper documentation in the Data Set Catalogs have been made into digital images, and then collected into a single PDF file for each Data Set Catalog. The inventory information in these DSCs is current as of July 1, 2004. This inventory information is now no longer maintained in the DSCs, but is now managed in the inventory part of the NSSDC information system. The information existing in the DSCs is now not needed for locating the data files, but we did not remove that inventory information.

The offline tape datasets have now been migrated from the original magnetic tape to Archival Information Packages (AIP's).

A prior restoration may have been done on data sets, if a requestor of this data set has questions; they should send an inquiry to the request office to see if additional information exists.

### 2. ERRATA/CHANGE LOG:

NOTE: Changes are made in a text box, and will show up that way when displayed on screen with a PDF reader.

### When printing, special settings may be required to make the text box appear on the printed output.

Version	Date	Person	Page	Description of Change
01				
02				

### 3 LINKS TO RELEVANT INFORMATION IN THE ONLINE NSSDC INFORMATION SYSTEM:

http://nssdc.gsfc.nasa.gov/nmc/

[NOTE: This link will take you to the main page of the NSSDC Master Catalog. There you will be able to perform searches to find additional information]

### 4. CATALOG MATERIALS:

a. Associated Documents

To find associated documents you will need to know the document ID number and then click here.

<a href="http://nssdcftp.gsfc.nasa.gov/miscellaneous/documents/">http://nssdcftp.gsfc.nasa.gov/miscellaneous/documents/</a>

b. Core Catalog Materials

### MULT-COORD SYS EPHEM & B-MODEL TAPE

### 64-060A-00G

THIS DATA SET HAS BEEN RESTORED. THERE WAS ORIGINALLY ONE
7-TRACK, 800 BPI TAPE WRITTEN IN BINARY. THERE IS ONE RESTORED
TAPE, WHICH WAS PACKED DURING THE RESTORATION PROCESS. THE DR
TAPE IS A 3480 CARTRIDGE AND THE DS TAPE IS 9-TRACK, 6250 BPI.
THE ORIGINAL TAPE WAS CREATED ON AN IBM 7094 COMPUTER AND WAS
RESTORED ON AN IBM 9021 COMPUTER. THE DR AND DS NUMBER ALONG
WITH THE CORRESPONDING D NUMBER AND TIME SPAN ARE AS FOLLOWS:

DR# DS# D# FILES TIME SPAN

DR005847 DS005847 D001796 1 10/04/64 - 09/29/65

### 5.46 MIN AVERAGE OF MAGNETIC FIELD

### 64-060A-02A

THIS DATASET HAS BEEN RESTORED. THERE WERE ORIGINALLY 5
9-TRACK, 800 BPI TAPES WRITTEN IN BINARY. THERE IS ONE RESTORED
TAPE. THE DR TAPE IS A 3480 CARTRIDGE AND THE DS TAPE IS
9-TRACK, 6250 BPI. THE TAPES WERE CREATED ON AN IBM 7094 COMPUTER.
THE DR AND DS NUMBERS ALONG WITH THE CORRESPONDING D NUMBERS AND
TIME SPANS ARE AS FOLLOWS:

DR#	DS#	DD#	FILES	TIME SPAN
DR02692	DS02692	DD 00237 DD 00238 DD 00239 DD 00240 DD 00241	1 2 3 4 5	10/04/64 - 11/02/64 11/02/64 - 11/30/64 12/20/64 - 01/18/64 01/19/65 - 02/09/65 03/07/65 - 04/05/65

### 5.46 MIN AVG BLCKD BCD VRSN OF 02A 5-MIN AVG C.R. CNT RATE SUMMARY TAPE

64-060A-02C

64-060A-03F

THESE DATA SETS HAVE BEEN RESTORED. THERE WAS ORIGINALLY ONE

7-TRACK, 556 BPI TAPE AND ONE 7-TRACK, 800 BPI TAPE, WRITTEN IN BCD.

THERE IS ONE RESTORED TAPE, WRITTEN IN EBCDIC. THE DR TAPE IS A

3480 CARTRIDGE AND THE DS TAPE IS 9-TRACK, 6250 BPI. THE ORIGINAL

TAPES WERE CREATED ON A 7096 COMPUTER. THE DR AND DS NUMBER ALONG

WITH THE CORRESPONDING D NUMBERS AND TIME SPANS ARE AS FOLLOWS:

DR#	DS#	D#	FILES	TIME SPAN
DR003454	DS003454	D006937 D000668	1-129 130	10/04/64 - 04/08/65 (03F) 10/05/64 - 04/04/65 (02C)

### MERGED MAGNETOMETER + EPHEMERIS

### 64-060A-02D

THIS DATA SET HAS BEEN RESTORED. THERE WAS ORIGINALLY ONE
7-TRACK, 800 BPI TAPE WRITTEN IN BINARY. THERE IS ONE RESTORED
TAPE. THE DR TAPE IS A 3480 CARTRIDGE AND THE DS TAPE IS 9-TRACK,
6250 BPI. THE ORIGINAL TAPE WAS CREATED ON AN IBM 7094 COMPUTER
AND WAS RESTORED ON AN IBM 9021 COMPUTER. THE DR AND DS NUMBER
ALONG WITH THE CORRESPONDING D NUMBER AND TIME SPAN ARE AS
FOLLOWS:

DR#	DS#	D#	FILES	TIME	SPAN
DR005226	DS005226	D002892	1	10/04/64 -	- 04/05/65

### 5.46MIN AV BLCKD BIN VRSN OF 02A

64-060A-02E

SPHE

00715

This data set has been restored. There were originally two 7-track, 800 BPI tapes written in Binary. There is one restored tape. The DR tape is a 3480 cartridge and the DS tape is 9-track, 6250 BPI. The original tapes were created on a 7094 computer and the restored tapes were created on an IBM 9021 computer. The DR and DS numbers along with the corresponding D numbers are as follows:

DR#	DS#	D#	FILES	TIME SPAN
			****	
DR005295	DS005295	D002899	1	10/04/64 - 01/18/65
		D002900	2	01/19/65 - 04/05/65

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- I. Identification
  - A. Spacecraft Description
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- II. Data Processing
  - A. 64-060A-00G
    - 1. Format of Ephemeris Tape (Input for merging Magnetometer and Ephemeris)
    - 2. Partial dump of Ephemeris tapes.
  - B. 64-060A-02A
    - 1. Format

    - Partial dump of output 7 track, 800 BPI, 7094 tape.
       Partial dump of input 9 track, 800 BPI, 360 tapes.
       Program listing used for conversion from 360 tape to 7094.
  - C. 64-060A-02C
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    - 2. Partial listing
  - D. 64-060-02D
    - 1. Format
    - 2. Partial Octal dump of tape D2892

    - Listing of time gaps
       Lable of printout of tapes
  - E. 64-060A-02E
    - 1. Format
    - 2. Partial Octal dump of tapes D-2897 and D-2900
    - 3. Listing of magnetometer 5.46 minute average.

### 52AC C BOLGF DESCRIPTION4

Explorer 21 (IMP 2) is a 135 pound, spin-stabilized spacecraft ingle lented for interplanetary studies of cosmic rays, magnetic fields, and incomes. It was launched into an elliptical orbit on October 4, 1964, with appropriate 15.9 earth radii (half the planned value) near local moon. Spacecraft performance was normal for most of the first four months, and for the sixth month, after leunch. Otherwise the data transmission was interminent, with final transmission occurring on October 13, 1965.

roject Nanager/Mr. Paul Butler/NASA-GSFC/Code 724
Project Scientist/Dr. Frank B. McDonald/NASA-GSFC/Code 560
Data Processing/Mr. C. J. Creveling/NASA-GSFC/Code 560
Orbital C mputations/Mr. D. J. Stewart/NASA-GSFC/Code 552
Attitude Computations/Mr. E. J. Pyle/NASA-GSFC/Code 711

### COACT OBJECTIVES

Explorer 21 (IMP 2) was the second in the series of Interplanetary Notice of Platforms. The objectives of this series are: (1) to study in the 1 the radiation environment of cislumar space, and to monitor this region over a significant portion of the solar cycle; (2) to study the collection properties of the interplanetary magnetic field and its dynamical relationship with particle fluxes from the sun; (3) to develop a flar prediction capability for Apollo; and (5) to further the description of the series of the interplanetary magnetic field and its dynamical relatively inexpensive spin-stabilized spacecraft for interplanetary investigations.

### SPAC OFT FULL DESCRIPTION

### C nfiguration

The main body formed a prism 12 inches long with octagonal faces 28 inches across. The axis of symmetry (spin axis) passed through the center of the faces. The main appendages were: four solar paddles, each being 20" by 26", with their long axes perpendicular to the spacecraft spin exis; four 16" antennas; a Rb-vapor magnetometer boom, 67" long mounts parallel to the spin axis; two rluxgate magnetometer booms, each 69" long, mounted perpendicular to the spin axis.

### Hower System

Proof was supplied by P/N solar call arrays mounted on both sides of the four paddles, and by a silver-cadmium battery pack. The average power produced by the solar calls was between 43 and 66 watts, depending on the spin axis-sum angle. The battery capacity was 5 amperenturs at 14 volts. The average steady state spacecraft load was 37 watts; during each 82 second interval, a ten second, 42 watt power drain and two 50 watt transients (of a few milliseconds duration) occurred. An undervoltage syl caused all systems to be turned off when the main system voltage (normally 19.6 volts) fell below 12 volts; a timer would turn the power back to after about 7.5 hours.

### SPACECRAIT FULL DESCRIPTION

### Cuboard Propulsion

None

### Communications

Data were telemetered to ground using pulse frequency modulation (PFM) of a 4 watt transmitter operating near 136 mc. A complete encoder sampling pattern was repeated every 327.7 seconds. A pattern consisted of three normal sequences followed by a Rb-vapor magnetometer sequence, each sequence being 81.9 seconds long. Each normal sequence consisted of 265 tone bursts in the frequency range 312.5 to 937.5 cps.

Information was conveyed by the frequency of the tone. Digital channels used eight discrete frequencies to telemeter three bits per burst. Analog thannels used the full frequency range to represent the channel input voltage (0 to 5 volts). During each normal sequence a complete set of spacecraft performance parameters and data from all experiments except the Rb vapor magnetometer were telemetered. During the Rb vapor magnetometer sequence, the magnetometer output directly modulated the transmitter in the frequency range 20 to 1000 cps. A range and range-rate transponder, amplitude modulating the same transmitter, was included for tracking.

### attitude Control

The spacecraft was spin stabilized, with an initial spin rate of 14.6 rpm. There was no active attitude control system onboard.

### Attitude Sensors

A digital solar-aspect sensor measured the spin axis-sun angle and the time the sensor meridian spun past the sun.

### Command System

The spacecraft had a telemetry on/off command designed to be used only in the termination of the spacecraft mission. Range and range-rate functions could also be commanded.

### Onboard Data Processing

None

### SPACECRAFT PERFORMANCE

A malfunction of the third stage of the launch vehicle caused a number of sevual spacecraft parameters to deviate significantly from their pre-launch nominal values. First, the apogee achieved (15.9 earth radfi, near the local noon meridian) was less than half its nominal value; thus interplantary measurements taken on IMP 2 are not nearly as extensive as on IMPs 1 and 3. Second, the spin axis direction (right ascension = 41.40, declination 47.40 on October 4, 1964; right ascension - 760, declination = 220 on April 7, 1965) was initially 780 removed from its nominal prelaunch value; this associate attitude gave a wide range of incident sun angles which in turn caused overheating and a partial failure of the battery after two months operation. Third, the initial spin rate of 14.58 rpm (which decreased slightly and then increased to 18 rpm) was below the anticipated 23 rpm; in some cases,

### SPACECRAFT ORBITAL PARAMETERS

	10/4/64	4/4/65	10/3/65
Apogee (geocentric, km)	101,778	100,394	100,023
Parigee (altitude, km)	193	1031	1465
Pariod (minutes)	2097	2077	2077
Inclination (degrees)	33.50	35.47	36.04
Eccentricity	0.88	0.862	0.855
Apogee Local Time	12:00	24:00	12:00

(100,000 km = 15.7 Earth Radii)

Source: IMP 2 Flight Report, Frank A. Carr, NASA TN D-3353.

# NSSDC TECHNICAL REFERENCE FILE

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### SPACECRAFT PERFORMANCE

this affected experiments' data analysis.

Despite the foregoing problems, the spacecraft telemetry system and all of the experiments functioned normally for several months from the October 4, 1964 launch date. After 57 days of normal operation, the battery proved incopable of sustaining spacecraft operation within the shadow of the earth; by 63 days after launch (December 6, 1964) the battery had failed completely. Thereafter, the spacecraft operated only during periods of favorable incident sun angles. Under these conditions, the spacecraft transmitted 56 days of data during the 59-day interval from December 12, 1964 to February 9, 1965 and 28 days of data during the 32-day interval from March 5, 1965 to April 5, 1965. The spacecraft continued to transmit intermittently until October 13, 1965, when one minute of data was received. All data acquisition efforts ceased as of November 1, 1965.

Nature October 4, 1964 and April 5, 1965, apogee, perigee and orbital period changed from  $9.54 \times 10^4$  km, 193 km and 34.95 hours to  $9.40 \times 10^4$  km, 1081 km and 34.62 hours, respectively. During the same period, the spin axis-sum angle increased from  $130^\circ$  to  $147^\circ$  and then decreased to  $60^\circ$ .

A more complete discussion of the IMP 2 performance history is given in NASA TW D-3353, "Flight Report Interplanetary Monitoring Platform IMP-II (Explorer 21)", by Frank A. Carr (June 1966).

### EXPONE OF BRIDE DESCRIPTION

Each of two unlaxial fluxgate magnetometers, having dynamic ranges of +40 gammas, samples the magnetic rield 30 times within each of six 4.8 second intervals each 5.46 minutes. Detector sensitivities are +0.25 gamma, and digitization uncertainty is +0.40 gamma. A rubidium vapor magnetometer was used in calibrating the fluxgates, but did not produce an independently used data set. The fluxgates functioned normally throughout the useful life. the satellite; last useful data was acquired on April 5, 1965.

### PERSONNEL

Principal Investigator/Dr. Norman F. Ness/NASA-CSFC/Code 616 wastigator/Dr. Donald H. Fairfield/NASA-GSFC/Code 616

### EXPERIM AT OR ECTIVES

The fluxgate magnetometers were intended to delineate precisely the vector characteristics of the interplanetary magnetic field and of the outer regions of the magnetospheric magnetic field.

### EXPERIM NT FULL DESCRIPTION

Our fluxgate magnetometer (0.75" diameter, 3" long) was mounted on each of two booms extending 69" from the main spacecraft body. A rubidium vapor magnetometer was included, but did not produce useful data due to poor signal to noise ratio. This detector is discussed in JGR, 69, p. 3531, and will not be further discussed here. (NSSDC does not expect to receive the Rb-vapor magnetometer data.)

The sensor of each fluxgate magnetometer was a saturable magnetic core which was driven at 10 kc/s from positive to negative saturation. Any second harmonic signal generated was due to the presence either of an ambient field component along the axis of the element or to permanent magnetization of the core material. The voltage output represent, the discriminated second harmonic output which is calibrated to yield the field component parallel to the sensor axis, while the phase indicates the direction, parallel or antiparallel.

Each magnetometer is calibrated in-flight, through a combined use of the Rb-vept magnetometer data, for 0.25 second each time it is a red on.

The dynamic range of the fluxgates was +40 gamma, with a sansitivity of ±0.25 g. ma. Subsequent digitization by the use of "comb filters" on the ground od to uncertainties of ±0.4 gamma. Analysis of the data indicates mat, to within the uncertainties just mentioned, the fields measured are indeed the ambient magnetic fields and are not contaminated by the spacecraft magnetic field.

The magnetic field is sampled by a single detector thirty times within each of six 4.8 second intervals each 5.46 minutes. A vector field is obtained from the 30 data points of each 4.8 second interval.

### EXP RI SNY FULL DESCRIPTION

More detailed discussions of the instrumentation, spacecraft structure and magnetic cleanliness, data sampling, and the analytic means whereby vector information is gained from uniaxial fluxgate measurements, is given in Ness, Scearce and Seek, JGR, 69, 3631-69, 1964. Spin modulation was removed from the data by special techniques made necessary by the abnormally low spin rate of IMP 2; these are discussed in Fairfield and Ness, JGR, 72, 2379-2404, 1967.

### EXPERIMENT PERFORMANCE

The two fluxgate magnetometers performed normally during the useful lifetime of the spacecraft, and provided good data until April 5, 1965. The vapor magnetometer operated intermittently because of poor signal-to-noise ratio and did not produce an independently useful data set.

### 64-0604-02

HISSOC TECHNICAL REFERENCE FILE

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DIASACKADARIC DISTURBANCES OF THE TO IS-MINUTE PRINTED ON THE MACHETOPAJSE AND THEIR RELATION TO HOW SHOCK SPIKES ACRES. J. GELPHYS. RES., 73, PETT-23de, 1968. 61-060A-05 66-058A-01 69-060A-02 62-041A-03 \* MARTHER & PAGE 11/17/69 \* GETGER TJOR 65-042A-02 HEXPLOSER 21 PEXPLORER 20 C4-060A-07 65-04EA-02 PHAR INER 4 \*CLASSON ANALYSIS OF PLUSTONISMS IN THE INSPIRANTIARY MAGNETIC PIELD INTERNED OF 140-FF ACCESSO, MASS-CORFE, ACCESSOR MASS-CORFE ACCESSOR MASS-CORFE ACCESSOR. SOLAR MACNETIC PIELOS ACCESSION ORDERED CATALOG SINFERR ANET HACHITIC PO SCEOMAGNETIC FIELDS HOLAN LANGUETTE DISTURDANCES AND THE INFERPLANETARY MAGNETIC PIELD \*MAGNETO PAUSE \*FLUXGATE MAG 64-050A-02 \*EXPLUSER 21 64-060A-02 64-060A-D2 50-VO 50- 69 +PI DIEE 3 7 \*CL. ASSC2 +1 NO 2 MACNETIC PIE, D MEASUREMENTS WITH THE IMP 2 SATELLINY SCLAR MAGNETIC PATTERN EXTENDING FROM 40 N TO 35 S ACRESTS. J. GECRAYS. 455. 72, 2379-2402, MAY 1967. A02803. 4A54-65FC. X-612-56-530. NOT. 1966. G72902. J. GEGRAYS. 855., 72, 2376-2402, MAY 1967. AC2407, U. OF CA. 1F., TECH. RPT. NO. 9. APR. 1950. STREET LANET MAGNETIC PO A02337. U. D\* CA... SE2. 3. 155. 60. MAY, 1967. SATELLITE STUDIES OF THE EARTH'S MAGSETIC TAIL, AD 2621, PAYS, UP MISNET OSHIESE, 409-434, 1968. ACIG75. NASA-5552. 4-612-67-338. JULY 1567. INTERPLANETARY SECTIA STRUCTURE, 1962-1506 PEXPLOSER 19 BEAPLORER 21 PROPERTY PACE PARAMENT CO \* MAGNETCTATL 23-046A-02 64-J77A-02 53-24CA-02 10-A 25 0-30 SILCOX.J. M. 1 GW 14 with 2 HOWARO, R. 7.55.N.P. 4855. Y.P. \* £x fl. 65 14, 18, 21, 28, 33 Botzes Filtricto. D. n. 301576 FAIFFELD. D. H. 301333 PUHANKSNIKAS. 301574 AMELUSUALAS. SCIAVE MILCON. J. M. GOLTON KANASHIAN. 4. 64-9 cut-02 62-6511-01 SOROS DANC + CG-0774-52 \* JFR C72+53 \* JAFG62759 \* JEGONGOOR \* 801796 NESS ...... . WAGINER . \*JEXCOUGH 0 co 140 x 140 0 act, 19532 SCLASTC2 \*CLASSAL 4CE A 5342 693 160 233010 372165 011665 093 100 111465

## NSSDC TECHNICA, REFERENCE FR.F.

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*Jr811145.3	NO1140	MPS 1.2	#1-/TK/VI	15 N-110

### IMP-2, Explorer 21 64-060A

64-060A-00D	Ephermis Tapes 7 track 556
64-060A-00E	Merged Ephemeris Tapes 7 track 800
64-060A-00F	Solar Ecliptic Ephemeris Tapes 7 track 556
64-060A-00G	Merged Solar Ecliptic Ephemeris Tapes 7 track 800
64-060A-00A	Fluxgate Magnetometer (originals) 9 track 800
64-060A-00B	Packed bin mag Tape has been released from Files
64-060A-00C	Packed BCD Flux Mag Tape 7 track 556
64-060-00D	Merged Magnetometer and Ephemeris Tapes 7 track 800
64-060-00E	Blocked Flux Magnetometer Tapes 7 track 800

### IMP-2, EXPL NO OF DESCRIPTION SATELLITE D# TAPES TRACK OF ID. DATA 0-01522 SOLAR THRU ECLIPTIC EPHEMERIS V4-0604-00F 0-01527 BLOCKED EPHEMERIS 64-060A-DOG D-01796 0-00237 FLUX GATE THRU MAGNETOMETER 64-0604-02A D-00241 D-02899 BLOCKED 2 FLUX MAGNETOMETER 64-0609-02E D-02900 MERCED MAGNETOMETER 64-0604-02D D-02892 AND EPHEMERIS PACKED BCD 64-060A-02C D-00668 MAGNETOMETER 64-0604-028 2-00667

EMP-2, EXPL 21

D#	NO OF TAPES	TRACK	DEN.	MODE	REMARKS
01522 THRU 01527	6	7	556	BIN	
01796	,	7	800	BIN	酸
00237 THRU 00241	5	9	800	BIN	
02999	2	7	800	BIN	
02892	1	7	800	B/N	**
-00668	,	7	556	BCD	
00667	1	7	556	BIN	TAPE HAS BEEN REMOVED FROM FILES

Problems Encountered on Marging the IMP 1, 2, and 3

Magnetometer and Ephemeris Data

The tape listing was in error. Binary 7 track magnetometer tapes existed for just some of the data, and not for all of it as stated in the listing. Also one of the formats for the existing 7 track binary tapes did not match what was actually on the tapes. As a result, 7 track binary tapes were made on the 360-75 from the original 9 track binary tapes and pifted for compatibility with the 7094.

The ephemeris tapes were in 7094-7044 DC5 packed form and so they had to be unpacked for use. They were edited by removing any overlapping data in order to simplify the merge. While the unpacked tapes were being edited, some bad data were encountered. Upon trying to remake them from the originals, it was discovered that some of the original tapes were no good and that data had to be emitted from the merge.

### Input Binary Tape Format

IMP 1, 2, and 3 Ephomeris 640609-006

Word	Symbol	Description
**1	IYR	Year
**2	IDCY	Day count of year
**3	IHR	Hour (UT)
**4	IMIN	Minute (UT)
45	GLAT .	Geodetic latitude in degrees
*3	GLONG	Geodetic longitude in degrees
••	RLAT	Geomagnetic latitude of satellite in degrees
•8	RLONG	Geomagnetic longitude of satellite in degrees
5	RAD	Radial distance from earth in earth radii
10	GMLONG	Geomagnetic longitude of sub- solar point
n .	GMLAT	Geomagnetic latitude of sub- solar point
12	SUNA	Angle in degrees between probe spin axis and satellite sun vector
10	XSE	X solar ecliptic coordinate of satellite
16	YSE	Y solar ecliptic coordinate of satellite
:5	ZSE	Z solar colliptic coordinate of satellite

	10		
	17	BPER	Perpendicular component of field (calculated perpendicular to spin axis)
	18	BPAR	Parallel component of field (calculated parallel to spin axis)
	19	BSE	Magnetic field in solar celiptic coordinates
	-20	рні	Angle between X and Y com- ponent of field in solar ecliptic in degrees
	21	THETA	Angle between field and pro- jection on X-Y plane in solar ecliptic coordinates in degrees
	23	PSI	Angle between payload X-axis and BPER in degrees
•	•23	XSOLM	Solar magnetospheric X-coor- dinate
	*24	YSOLM	Solar magnetospheric Y-coor- dinate
24	-25	ZSOLM	Solar magnetospheric Z-coor- dinate
	26	FMAT (1,1)	Rotation matrix to go from payload coordinates to solar ecliptic
	27	FMAT (1,2)	
	28	FMAT (1,3)	
	29	FMAT (2,1)	
	30	FMAT (2,2)	
		FMAT (2,3)	
9	52	FMAT (3,1)	

Magnitude of magnetic field

33	FMAT (3, 2)	
34	FMAT (3,3)	
35	SESM (1,1)	Rotation matrix to go from solar celiptic to solar mag- netospheric coordinates
36	SESM (1,2)	
37	SESM (1,3)	
38	SESM (2,1)	
39	SESM (2,2)	
40	SESM (2,3)	
42	SESM (3,1)	The Alexandrea
-2	SESM (3,2)	
43	SESM (3, 3)	
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Written on output tape

\*\* Used to determine if ephemeris data matches magnetometer data

Each physical record is 216 words long containing five 43-word logical records and a FORTRAN control word at the beginning.

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FILE 0001 REC 01	001 0	1 1296	1964	Ост. 4		
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	0007	21424854931	202402533442	210552377534	605400542222	
		2147655CC000	214437200000	214633200000	214765505662	
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	0289	505637534121	204614631463	605571463146	207460314631	
	0337	575730365025	601563401203	601667213634	214576514631	2
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		210545236536	605400564632	210 40 231 4631	605575463146	5
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	0573	200514212066	000000000000	600504155162	604466043220	5
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	0817	505704631463	207512631463	202533110562	000426000000	00
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FILE COCI DEC 0003 CH	605772702436	205665727 024	6 06401463146	207544314631	202
	200606435024	601744471 075	602524473022	213416214631	212
	6 0544 0 256 977	211445063146	200606435024	602551241473	60
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	000327000001		21 052 044 406 6	604772332465	21
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	201656211573	602422613672	201656211573	602661217337	60
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	210513311331	604 76567 0202	211440615062	605474617615	21
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	7 6 0 6 4 3 1 4 6 3 1 4 6	207574314631	211640314631	206677146314	2
	5 502712111353	211644063146		200614067645	01
	3 202404102744	602711477752	200602101233	200404077060	2
	600452564466	177652047352		577703753677	2
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	7 210476364234	604752667376		202530443122	6

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02314631	505454745254	211446631463	200742763365	602572227241	50 167 60 40 1 36
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0 21 34 74 3	200404077060	201400000000	000000000000	00000000000	000000000000
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51631453	201522006542	602627604324	601756003505	200614125577	0000000000000
52 01 3341 7	500452534222	177651727220	200602127326	200404077060	201400000000
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156314631	21245 360 1133	211434431722	605474744570	211453146314	201601130756
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556211573	602661217337	602412775501	200614106621	000000000000	000000000000
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		(4)			10	5 0		1.958	-4.238	-1.572	2.443	2.625	1.375
	280		995		10	5 0		1.200	-1.042	4.128	0.770	1.677	0.757
	280				10	5 0			-4.00B	-1 . 698	1.154	2.555	1.171
-	580	1700			10	5 1		3.644	-1.409	-2.293	1.167	1.775	2.887
	280				10		11.4	1, 209	-2.715	-1.672	1.541	3.138	2.736
	280	18 2		54	10	0 1		1.518	-3.405	2.092	0.989	0.676	1.439
-	280		0.35		10	25	25.3	1.958	-3.768	2.781	0.436	0.540	0.562
	280	18 2	C39	64	10		21.0	2.667	-0.272	1.053	1.273	1.207	2.270
	2 E C	18 3		54	10		2 0.5	2.294	1.864	1.659	0.837	1.348	0.773
	280	18 2	067	5.4	10	6	2 6.0	2.812	-3.040	2.076	0.746	0.906	1.350
	280	18 2	071	04	10		2 11.5	2.409	-2.840	2.155	0.811	1.154	1.177
	280	18 2	075	40	10		2 15 00	1.471		3.615	1.261	0.519	0.815
-	280	18 2	e79	94	10		2 22.4	1. 911	-0.858	2.319	1.981	2.622	C.804
	280	18 2	103	64	10		2 55.1	- ve 051	-1.843	4.576	0.645	0.719	0.300
	280	18 2	107	04	10	6	3 0.5	-0.246	-0.086	3.467	1.535	1.243	2.361
	200	18 2	111	54	10	0	3 0.3	-1.158	-1.106	2.233	1.065	1.060	1.355
	580	18 2	1115	64	10	6	3 11.5	-0.159	-3.154		0.343	0.835	0.322
	280	18 2	1114	64	10	0	3 10.9	2.019	-1.675	3.823	0.852	0.717	0.823
	280	18 2	2143	04	10	. 6	3 49.7	1.519	-3 +6 68		0.488	C.211	0.413
	280	18 2	147	64	10	0	3 33.1	0.934	-4.132	2.303	0.598	1.315	1.844
	380		2151	69	10	6	4 0.0	0.574	-2.985	-2.222	1.045	0.901	0.691
	280			54	10	6	4 0.0	1.059	-4.622	-0.157		0.388	0.984
	280		2159	54	10	6	4 11.5	0.571	-4.537	0.919	0.854	0.492	0.634
	280	-	£ 183	64	10	6	4 44.2	0.638	-3.802	2.307	0.520	1.106	2.402
	280		2187	54	10	6	0 47.7	-0.179	-4.401	0.269	1.196	0.983	1.486
	280	-	2191		10	6	4 55.1	3. 023	-1.261	1.513	1.123		1.309
	280		2195			0	5 0.0	2.594	-1 -1 98	1.568	1.213	1.268	1.451
	280		2199		1 (		5 5.1	0.682	-2.263	2.802	1.265	0.972	0.739
-	280		2223				5 38.8	306 0	-3.490	-1.500	0.571		0.901
	280		2227		100		5 44.3	1.744	-2.605	-1.345	0.633	1.246	1.034
	280		1525	04			5 44.7	2.486	-1.808	1.555	0.796	1.387	1.473
	280		2235		1.4		5 55.2	2.964	-2.018	0.701	1.557	1.538	- PER . PERF 22
			2235		Paulo		6 0.6	3. 365	-1.047	-0.177	1.075	1.239	1.240
	280		2203				0 33.4	3.427	-0.193	1.4 34	0.933	0.567	1.112
-	280	1023800	2267				6 33.8	2.546	-2.747	1.348	0.278	0.707	0.737
	280	-	2271	69	1000		5 +4 .3	2.399	-0.793	0.244	1.088	1.430	1.932
1	280		2275				6 49.7	2. 787	-1.026	0.365	0.644	0.784	0.904
-	560						6 55.2	0.472	-2.054	-1.736	2.339	4.767	4.383
9	280		2279				7 27.9	-0.255	-2.716	0.905	1.906	0.863	1.999
1	280		2303				7 33.4	1.036	-3.117	-0.132	0.	G.	0.
-	280		2307				7 30.8	2.047	-1.491	1.614	1.656	1.515	
	560		2211				1 44.3	-2.878	-3.599	-0.970	0.256		0.991
SW	286	16	2315	3	. 1		7 49.5	-2.024	-3.868	0.504	1.087		1.613
1 -	280	16	2319	0.	· 1		H 22.5	-1.914	-3.873	-1 .486	0.625	0.555	0.357
	586		2343					1. 031	-3.561	1.617	1.341	1.339	1.045
	580		2347				B 33.4	1.088	-3.960	-0.015	1.039		-
1 -	280	16	2351	0	. 1		8 33.9	2.124	-3.088	0.773	1.082	0.926	The second secon
	580	16	2356	0.04	4 1			1.076	0.073	4.617	0.570	0.197	0.298
-	590	15	2305	1 .67	. 1	0 0	3 44.3						UELLI
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	GED DAT					SP ACECRAFT	
TELD	VARIAN		FIELD	FIELD	FIELD	RADIAL	G000
S.E.	Y	Z	MAGNITUDE	THETA	PHI	DISTANCE	POINTS
X		-					
533	0.261	0.969	4.623	-8.655	294.996	13.744	:5
537	1.543	1.267	5.354	-1.697	31 5 . 831	13.787	12
853	1.810	1.736	4.364	-5.001	294.380	13.831	12
443	2.625	1.375	4.926	-18-612	294.801	13.884	12
770	1.677	0.757	4.426	68.869	319.220	13.574	12
144	2.555	1.171	4.400	-22.694	279.129	14.233	15
167	1.775	2.887	2.948	-51.066	310.748	14.279	12
541	3.138	2.736	3.531	-28.258	259.215	14. 324	12
989	0.676	1.039	4.450	28.046	299 .902	14.356	12
436	0.540	0.562	5.390	31.068	305.291	14.401	12
273	1.207	2.270	2.538	24.501	353,236	14.663	12
837	1.348	0.773	3.764	26.322	33.545	14. 700	12
746	0.906	1.350	4.399	28.151	308.395	14.733	12
811	1.159	1.177	3.856	33.970	257.390	14.771	12
261	0.519	0.815	4.167	59.714	334.838	14.804	12
981	2.522	0.804	2.965	51.440	268.415	15.018	12
645	0.719	0.300	4.584	86.738	199.156	15.056	13
.535	1.243	2.361	3.819	65.209	223.670	15.081	12
. 065	1.060	1.355	3.868	35.263	267.117	15.100	12
.343	0.835	0.322	4.636	55.543	320.328	15. 139	12
852	0.717	0.923	4.588	30.072	292.497	15.315	12
.488	0.231	0.413	4.822	28.526	282.742	15.341	12
598	1,316	1.844	3.765	-36.169	280.889	15.367	12
.045	0.901	0.691	4.745	-2.014	282.902	15. 392	12
. 854	0.388	0.984	4.664	11.353	277.169	15.404	12
.520	0.432	0.634	4.493	30.899	279.529	15.550	8
.196	1.106	2.402	4.413	3.492	267.671	15.567	12
.123	0.983	1.486	3.608	24.800	337.363	15.565	12
.213	1.268	1.309	3.577	26 . 0 10	323.813	15.598	12
.265	0.972	1.451	3.666	49.849		15.624	12
.571	0.570	0.739	3.90€	-22.582		40 330	12
. 633	1.246	0.901		-23.232	303.809		12
. 796	1.387	1.034	3.444	26 .8 36	323.968	15,744	
.667	1.588	1.479	3.653	11.062		15.758	12
.075	1.239	1.240		-2.696			12
.933	0.567	1.112	3.720	22.671	356.783		4
. 278	0.707	0.737		18.813			12
. 088	1.430	1.932		5.516			11
.644	0.784	0.994	2.992	7.002			The second second
. 339	4.767	4.383		-32.783	280.091	15. 665	4
. 906	0.863	1.999	2.874	18.367	264.629	The second secon	1
	C.	0.	3.520	-2.083			12
.556	1.515	0.922		32.513	717		12
. 266		0.991		-11.893	The second secon		12
.087		1.013		6.588			12
2.625		0.357		-18.956			12
.341	1.339	1.045		35.430	Section 1971		12
1.039	1.384	1.590		-0.212			12
1.082		1.727		11.654			11
AND RESIDENCE OF THE PARTY OF T	0.197	0.298	4.742	76.852	3.899	15.866	

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